A note on the numerical treatment of the k-epsilon turbulence model *

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Abstract

Numerical solution of the equations arising from the \( \kappa - \epsilon \) turbulence model has difficulties inherent to nonlinear convection-reaction-diffusion equations with strong reaction terms, resulting in that numerical schemes easily become unstable. We present a formulation that stresses on the robustness of the solution method, tackling common problems that produce instability. The main contribution concerns the loss of positivity of \( \kappa \) and \( \epsilon \), which is addressed by acting on the coefficients of the reaction and diffusion terms rather than on the turbulent variables themselves. In addition, a linearized implicit, non-iterative, treatment of the wall law is proposed.

Keywords: turbulence, numerical methods, k-epsilon model, finite elements

1 Introduction

The numerical simulation of turbulent flows modeled by the \( \kappa - \epsilon \) model has been the subject of much research in the last years. This article is mainly focused on the stability of numerical approximations for this model. Though mathematical results exist ensuring the well-posedness of the equations [7], the strong nonlinearities may interact with discretization errors in such a way as to instabilize computations. This indeed happens if, as frequently occurs, no physically meaningful initial condition is at hand.

A typical behavior of unstable computations involves the loss of positivity of \( \kappa \) or \( \epsilon \). This changes the sign of several terms in the equations, with disastrous effects. To avoid such loss of positivity is not an easy task. One may limit \( \kappa \) and \( \epsilon \) from below (e.g., [5]) but the solution strongly depends on the limitation strategy and numerical results are frequently unacceptable. The failure of this simple approach motivates the use of numerical techniques that preserve positivity [3, 7], but in most unstructured meshes obtuse angles exist that preclude these algorithms from guaranteeing positive results. Though elaborate iterative techniques provide further improvement [3], the difficulty is not removed. Another alternative is to solve for the logarithms of \( \kappa \) and \( \epsilon \) [4], but one then deals with a modified set of equations involving exponentials of the unknowns.

The numerical trick we propose here is very simple and can be implemented in most \( \kappa - \epsilon \) solvers. The idea is to look at the linearized equations for \( \kappa \) and \( \epsilon \) as convection-diffusion-reaction equations, and to notice that their coefficients are always positive for the exact solution. Since a negative coefficient must be due to discretization errors, it can be put to zero remorselessly. One never touches the values of \( \kappa \) or \( \epsilon \), just the coefficients of the linearized equations. This trick, together with an improved treatment of the wall-law, results in a very robust scheme. It was combined with a recent equal-order method introduced by Codina and Blasco [2], using linear interpolation for all the unknowns. The following sections describe the proposed method, with numerical tests assessing both its stability and accuracy.

2 Problem Formulation

Let \( \bar{u}, p, f, \rho \) and \( \mu_t \) be the Reynolds-averaged velocity, pressure, volumetric force, density and viscosity of the fluid, respectively, and also let \( \nu_t = \mu_t/\rho \). We start from the Reynolds Averaged Navier-Stokes (RANS) equations for

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incompressible flow
\[
\rho \frac{\partial \vec{U}}{\partial t} + \rho \vec{U} \cdot \nabla \vec{U} + \nabla p - \mu \Delta \vec{U} + \rho \nabla \cdot \vec{R} = f
\]
\[
\nabla \cdot \vec{U} = 0
\]
(1)
(2)

Here \( R \) is the Reynolds Stress Tensor for which the \( \kappa - \epsilon \) turbulence model proposes
\[
R = -\frac{2}{3} \kappa \mathbf{I} + \nu_T \left( \nabla \vec{U} + \nabla \vec{U}^T \right), \quad \nu_T = c_\mu \frac{\kappa^2}{\epsilon}
\]
(3)

where \( c_\mu = 0.09 \), \( \kappa \) is the turbulent kinetic energy and \( \epsilon \) is the dissipation of \( \kappa \). Replacing \( R \) from (3) in the RANS equations we obtain equations identical to the laminar case but with an effective viscosity \( \nu_{ed} = \nu_\ell + \nu_T \) and an effective pressure \( P = p + \frac{\kappa}{\kappa} \).

The equations for \( \kappa \) and \( \epsilon \) are of the convection-diffusion-reaction type, with non-linear coefficients, as given below in a form suitable for later use [7]:
\[
\frac{\partial \kappa}{\partial t} + \vec{U} \cdot \nabla \kappa - \nabla \cdot (D_\kappa \nabla \kappa) + \gamma_\kappa \kappa = F_\kappa
\]
(4)
\[
\frac{\partial \epsilon}{\partial t} + \vec{U} \cdot \nabla \epsilon - \nabla \cdot (D_\epsilon \nabla \epsilon) + \gamma_\epsilon \epsilon = F_\epsilon
\]
(5)

where the diffusion coefficients are given by
\[
D_\kappa = \frac{\nu_T}{\sigma_\kappa} + \nu_\ell, \quad D_\epsilon = \frac{\nu_T}{\sigma_\epsilon} + \nu_\ell
\]
(6)

the reaction coefficients by
\[
\gamma_\kappa = \frac{\epsilon}{\kappa}, \quad \gamma_\epsilon = \frac{c_2 \epsilon}{\kappa}
\]
(7)

and the source terms by
\[
F_\kappa = \frac{\nu_T}{2} \left| \nabla \vec{U} + \nabla \vec{U}^T \right|^2, \quad F_\epsilon = \frac{c_1 \kappa}{2} \left| \nabla \vec{U} + \nabla \vec{U}^T \right|^2
\]
(8)

with \( c_1 = 0.126 \), \( c_2 = 1.92 \), \( \sigma_\kappa = 1.0 \) and \( \sigma_\epsilon = 1.3 \). Notice that the diffusion and reaction coefficients, and the source terms, are non-negative for physically admissible solutions (non-negative \( \kappa \) and \( \epsilon \)).

The previous equations are defined on a domain \( \Omega \), that we assume bounded by \( \partial \Omega \). The portion of \( \partial \Omega \) corresponding to solid walls is denoted by \( \Gamma_w \), along which a layer of thickness \( \delta \) has been removed and replaced by a wall law. This amounts to the imposition of the no-penetration condition \( \vec{U} \cdot n = 0 \) (\( n \) is the normal), of a tangential stress \( \tau^w \) (opposite to local velocity) obeying
\[
\frac{|\vec{U}|}{\sqrt{\rho}} - \frac{1}{\chi} \ln \left( \frac{E \delta}{\nu_\ell \sqrt{\tau^w / \rho}} \right) = 0
\]
(9)

and of Dirichlet-type values for \( \kappa \) and \( \epsilon \)
\[
\kappa = \frac{u^*}{\sqrt{\nu_\ell}} \quad \epsilon = \frac{u^*}{\chi \delta}
\]
(10)

In (9)-(10), \( E = 9 \) for smooth walls, \( \chi = 0.41 \), \( u^* = \sqrt{|\tau^w / \rho|} \), and defining \( \delta^+ = u^* \delta / \nu_\ell \), the inequalities \( 20 \leq \delta^+ \leq 100 \) must hold.

It is customary to specify the value for \( \delta \) over \( \Gamma_w \), so that in order to find \( \tau^w \) one must solve the nonlinear equation (9). A second possibility has been successfully applied in this work: To specify the value of \( \delta^+ \) over the boundary with wall law condition. The main advantage of this choice is that given \( \delta^+ \) we can calculate \( \tau^w \) in a direct fashion (no iterations that could fail to converge). As a balancing drawback, now \( \delta \) depends on the solution, and can take different values at different points on the boundary. This is however not an issue when \( \delta \) is negligible compared to the dimensions of the domain.
3 Numerical Scheme

The implemented numerical scheme consists of four fractional steps: (1) Calculate the projection of the pressure gradient, (2) Solve continuity and momentum equations, (3) Solve equation for \( \kappa \), and (4) Solve equation for \( \varepsilon \). The first two will not be discussed here (see [2]), except for the treatment of the wall law that is deferred to paragraph 3.2. Let \( K_h \) and \( E_h \) be the interpolation spaces for \( \kappa \) and \( \varepsilon \), respectively, consisting of continuous, piecewise linear functions. The implemented variational formulations are:

**Fractional step 3:** Let \[ \frac{s^n + s^n + n}{\Delta t} + \sum_{K \in T_h} (R_{\kappa}, \tau_n \mathcal{P}_n(s_h))_K = 0 \quad \forall s_h \in K_h \tag{11} \]

**Fractional step 4:** Let \[ \frac{t^{n+1} + t^{n+1}}{\Delta t} + \sum_{K \in T_h} (R_{\varepsilon}, \tau_{\varepsilon} \mathcal{P}_\varepsilon(t_h))_K = 0 \quad \forall t_h \in E_h \tag{12} \]

where \((u, v) = \int \Omega u v d\Omega\) and spaces with a dot above indicate homogeneous Dirichlet data. The first two terms in Eq. (11) (and analogously for Eq. (12)) come from the Galerkin formulation. The third one is a stabilization term, involving an algorithmic parameter \( \tau_{\kappa} \) and a perturbation function \( \mathcal{P}_\kappa \), which introduces upwinding. Specific designs for this term are provided by the SUPG or SGS methods [1].

3.1 Positivity of the coefficients

The numerical scheme as explained so far performs poorly due to its inability to guarantee that \( \kappa \) and \( \varepsilon \) remain positive. Undershoots develop in boundary and internal layers, and negative values of either \( \kappa \) or \( \varepsilon \) result in the immediate destabilization of the scheme. However, by inspection of the formulation it is realized that it is not the negative values of \( \kappa \) and \( \varepsilon \) that really matter, but instead the appearance of negative reaction or diffusion coefficients that cause the exponential growth of the solution. We propose to limit \( \nu_T, \gamma_\kappa, \gamma_\varepsilon \), and \( F_\varepsilon \) from below without touching the nodal values of \( \kappa \) or \( \varepsilon \). The rationale for this is that we can accept local regions with negative values of these fields as long as these “problematic” regions remain localized and do not lead to global instability. The modifications introduced in the numerical scheme are:

- \( \nu_T \) is limited from below to a small fraction of \( \nu_\varepsilon \), and evaluated at the previous time step.
  \[ \nu_T = \max\{c_\mu \frac{(\kappa^n)^2}{\varepsilon}, 10^{-4} \nu_\varepsilon\} \tag{13} \]

- The rest of the coefficients are limited from below by zero,
  \[ \gamma_\kappa = \max\{c_\nu \kappa^n, 0\} \tag{14} \]
  \[ \gamma_\varepsilon = \max\{c_2 \varepsilon^n, 0\} \tag{15} \]
  \[ F_\varepsilon = \max\{c_1 \kappa^{n+1} - \frac{1}{2} \left| \nabla \bar{U} + \nabla \bar{U}^T \right|^2, 0\} \tag{16} \]

3.2 Implementation of the wall law

From (9), \( \tau^w \) can be written as
\[ \tau^w = -g \left( \left| \bar{U} \right| \right) \frac{\bar{U}}{\left| \bar{U} \right|} \tag{17} \]
where the function \( g \) simply equals \( \rho \left[ \chi \left| \bar{U} \right| / \ln(E \delta^+) \right]^2 \) when \( \delta^+ \) is specified, but is an implicit expression (iterations needed) if \( \delta \) is specified. All of the following can be done in either case, the latter requiring differentiation of implicit functions.
Evaluating $\tau^w$ at the previous time step leads to instability. We use instead a linearized implicit scheme, i.e.,

$$\tau^w_i \left( \bar{U}^{n+1} \right) \approx \tau^w_i \left( \bar{U}^n \right) + \frac{\partial \tau^w_i}{\partial \bar{U}_j} \bigg|_{\bar{U}^n=\bar{U}_j} \left( U_j^{n+1} - \bar{U}_j^n \right)$$

(18)

where summation over repeated indices holds. Using (17) and some algebra we get

$$\tau^w_i \left( \bar{U}^{n+1} \right) = -g \left( |\bar{U}^n| \right) \left( \frac{U_j^{n+1}}{|U^m|} - \frac{U_j^n U_j^{n+1}}{|U^n|^2} \right) - \frac{\partial \mu}{\partial |\bar{U}|} \left( \frac{U^n U_j^n U_j^{n+1}}{|U^n|^2} - U_i^n \right)$$

(19)

This formulation turns out to be stable. Furthermore, the terms multiplying $\bar{U}^{n+1}$ in (19) lead to a positive contribution to the system matrix.

4 Numerical results for two test problems

4.1 Fully developed flow in a channel

This test consists of the flow between parallel plates due to a constant pressure gradient. The following flow conditions were used: $\partial P/\partial x = -0.52$, $\nu_l = 10^{-4}$, $H = 1$, resulting in a Reynolds number $Re_H = 120,000$, based on $H$, the channel height, and $U_0$, the maximum velocity. Wall-law boundary conditions were imposed, fixing $\delta = 0.01196H$.

Several numerical tests were performed, those chosen as representative are listed in Table 1. Initial condition A corresponds to $U$, $\kappa$ and $\epsilon$ set to zero at $t = 0$. This initial condition is physically meaningful. Conversely, initial condition B corresponds to $U=100$, $\kappa=0$, $\epsilon=0$, which is non-physical and generates a strong initial transient. Case 1 addresses the accuracy of the method, while cases 2 to 4 were used to test its robustness.

In cases 1, 2 and 4 a steady state was reached, and case 3 oscillates. The solutions obtained for the steady state, not shown here for brevity, agree with the experimental results by Laufer and the numerical results by Utnes [9]. Also, the solutions obtained with different grids and time steps (cases 1, 2 and 4) practically coincide.

The usefulness of the proposed tricks is realized looking at Figs. 1 and 2, where the maxima and minima of $\kappa$ and $\epsilon$ during the transient are shown. Wild oscillations appear (locally!) in cases 2, 3 and 4. However, in cases 2 and 4 the steady state is reached without problem. Moreover, persistent oscillations such as those in case 3 can be avoided by refining either the temporal or spatial discretizations (leading to cases 2 and 4, respectively).

4.2 Backward facing step (BFS)

The geometry of this problem (2D) is shown in Fig.3. The dimensions are $H = 1$, $H_1 = 2$, $L_1 = 14$ and $L_2 = 30$. Wall laws (imposing $\delta^+ = 30$) hold along top and bottom boundaries. Inflow conditions correspond to fully-developed flow. The case discussed here corresponds to $Re = 44479$ (based on step height $H$ and maximum inlet velocity $U_0$), with $\nu_l = 2.88 \times 10^{-4}$ and $\rho$ = 1.0, which lead to $U_0 = 12.81$. The initial condition is uniform over $\Omega$, $U = 1.0$, $P = 0.0$, $\kappa = 0.01$, $\epsilon = 0.01$. Notice that this does not respect mass conservation and is far away from the steady state solution. The grid consists of 6823 nodes and 13280 linear triangles (see Fig. 4). With a quite large time step ($\Delta t = 0.3125$) a quasi-steady state was obtained without difficulty in 240 steps. Nevertheless, at a few nodes near corner a the turbulent dissipation $\epsilon$ kept oscillating. This local phenomenon did not pollute the solution away from this corner, and could be damped by further reduction of the time step (100 additional steps with $\Delta t = 0.003$ proved sufficient).

Though not presented here, comparison of velocity and Reynolds stress profiles with previous experimental and numerical results [10] is satisfactory (see [6] for details). The obtained reattachment length, frequently used for comparison, was $L_r = 6.59$, in good agreement with both an experimental value $L_r \approx 7$ and several works reporting that the $\kappa - \epsilon$ model underestimates $L_r$ by 10% to 25% [8].

Coming back to the central topics of this article, there are regions of the flow in which $\kappa$ or $\epsilon$ are negative, even at steady state. It must be recalled that our method allows for the appearance of such regions, and that our speculation is that in physically meaningful transients or at the steady state they will be sharply localized and will not affect the overall solution. For the present example at steady state (things are much worse during the initial transient) such regions are shown in Fig. 5. It is clear that $\kappa$ or $\epsilon$ are negative at just a few nodes. Furthermore, this happens where crossflow gradients are steep, and could be avoided by refining the grid or by introducing a discontinuity-capturing operator in the formulation. The minimum values of $\kappa$ and $\epsilon$ (see Table 2) are indeed small (in modulus) as compared to the maximum values.
5 Concluding remarks

A robust algorithm to solve the $\kappa - \epsilon$ equations has been presented. Its design is based on the manipulation of the different terms so as to add as many positive terms as possible to the system matrix and to keep source terms also positive. Then, as discretization errors may induce non-physical negative values of diffusion or reaction coefficients, these are limited from below by zero. This approach works much better than the frequently used limiters on the turbulent variables. Along the same design line is the proposed treatment of the wall boundary condition by semi-implicit linearization, an additional trick being the imposition of $\delta^+$ instead of $\delta$, avoiding iterations in the wall-law algorithm that may fail to converge. Remarkably, the proposed algorithm can easily be implemented in most flow solvers, not just finite element ones. Also worth mentioning is the good behavior of the recent stabilized method of Codina and Blasco [2], for which few reported experiments (none of them turbulent) exist.

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References


<table>
<thead>
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<th>Case</th>
<th>Number of nodes (N)</th>
<th>Δt</th>
<th>initial condition</th>
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<tr>
<td>1</td>
<td>260</td>
<td>0.01</td>
<td>A</td>
</tr>
<tr>
<td>2</td>
<td>31</td>
<td>0.01</td>
<td>B</td>
</tr>
<tr>
<td>3</td>
<td>31</td>
<td>0.1</td>
<td>B</td>
</tr>
<tr>
<td>4</td>
<td>260</td>
<td>0.1</td>
<td>B</td>
</tr>
</tbody>
</table>

Table 1: Details of the numerical tests for the channel flow problem. The grids are nonuniform, refined near the walls.

<table>
<thead>
<tr>
<th>max κ</th>
<th>6.5305</th>
</tr>
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<tbody>
<tr>
<td>min κ</td>
<td>-0.0708</td>
</tr>
<tr>
<td>max ε</td>
<td>922.924</td>
</tr>
<tr>
<td>min ε</td>
<td>-20.4855</td>
</tr>
</tbody>
</table>

Table 2: Maximum and minimum values of κ and ε at the steady state of the BFS test problem.
Figure 1: Channel flow problem. Evolution of the maximum (a) and minimum (b) of $\kappa$ in the initial transient for each of the cases.
Figure 2: Channel flow problem. Evolution of the maximum (a) and minimum (b) of $\epsilon$ in the initial transient for each of the cases.
Figure 3: BFS problem. Geometry and dimensions of the computational domain.

Figure 4: BFS problem. Central portion of the grid
Figure 5: BFS problem. Regions with negative values of $\kappa$ (a) and $\epsilon$ (b).